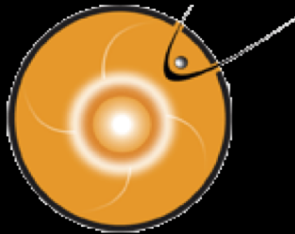


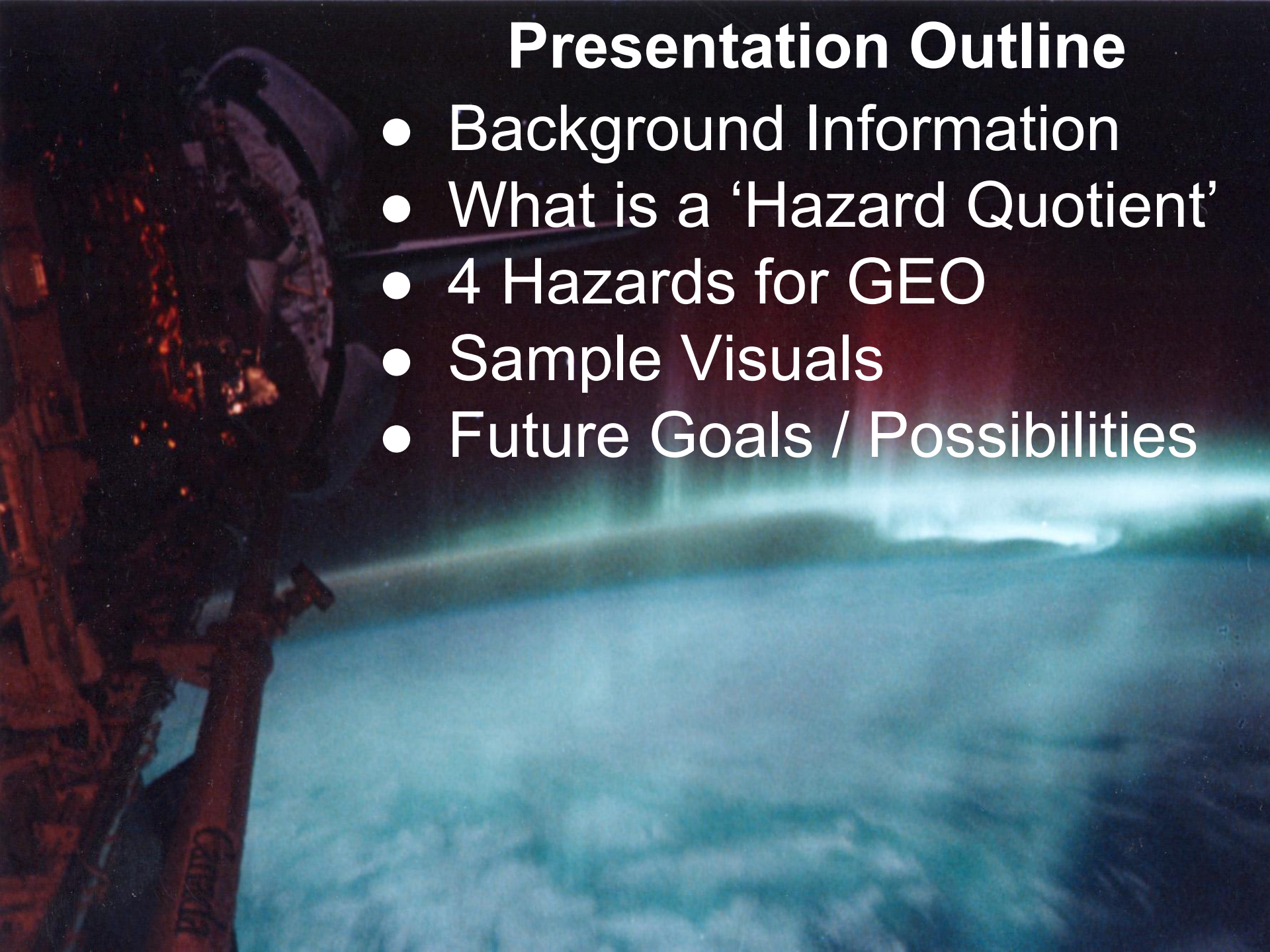
Development of Spacecraft Environmental Anomalies Expert System (SEAES) at NASA



Dhanesh Krishnarao

Presentation Outline

- Background Information
- What is a 'Hazard Quotient'
- 4 Hazards for GEO
- Sample Visuals
- Future Goals / Possibilities



Background Information

- Post Anomaly Analysis Tool
 - Way for spacecraft operators to be easily analyze and understand space environment around their satellite
 - Does not require technical knowledge of space weather
- Use Paul O'Brien's algorithm to determine a 'hazard quotient'
- User can set custom threshold for future automated watch and warnings

What is a 'Hazard Quotient'

- Hazard Quotient: The ratio of instantaneous to mission averaged likelihood of an anomaly for a specific space weather effect

Surface Charging

Single Event Effects

Internal Charging

Total Dose



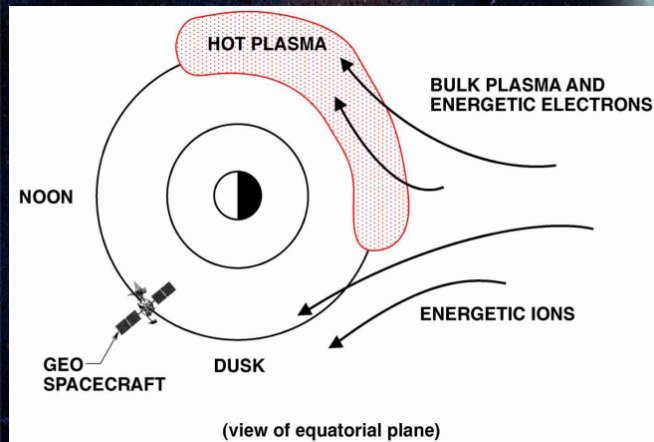
Surface Charging

- Surface Charging: Buildup of electric charge on the surface of a spacecraft structure

- Large Currents of Low Energy (~ 10 KeV)

- Using O'Brien's table of values

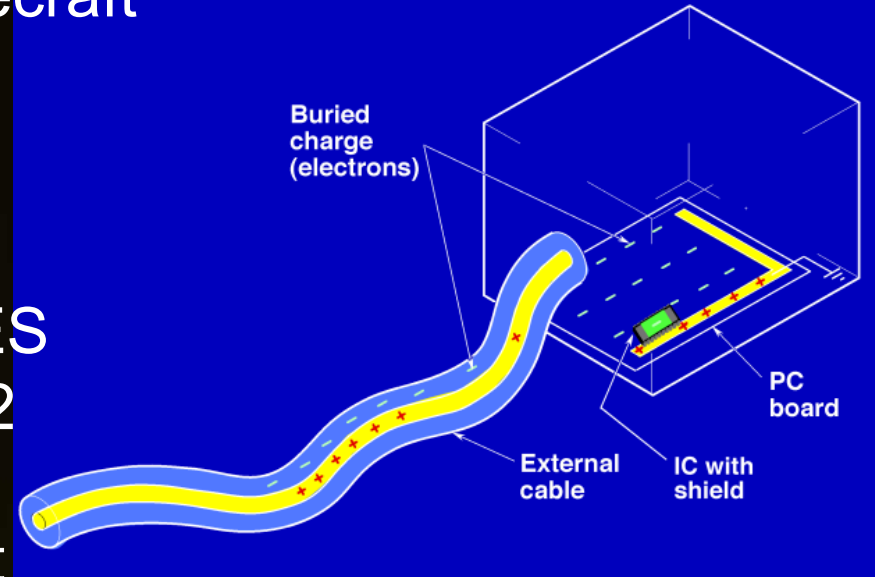
- Based on historical analysis of Surface Charging in relation to Kp and MLT of spacecraft.



- **Internal Charging:** Buildup of electric charge anywhere within a spacecraft structure

- Small currents of high energy electrons

- >2 MeV Electron Flux at GOES
 - averaged over previous 12 hours
- Local Time (LT) at Spacecraft




- Asymmetric electron radiation belt
 - Use two different methods to approximate the Flux at different longitudes.
 - O'Brien (2009)
 - Yi-Jiun Su (2014)

Internal Charging

Single Event Effects

- **Single Event Effects**: Solar Energetic Particles causing single upsets (bit flips, etc.). Charge deposited inside integrated circuits
- Primarily caused by MeV protons
 - Use GOES >30 MeV Proton Flux
- May also be from heavier nuclei



- 
- Total Dose: Coming from SEP events, a large amount of energy can build up on solar arrays
 - Susceptible to protons with Energy ~ 5 MeV
 - Use GOES >5 MeV Proton Flux
 - averaged over previous 24 hours

Total Dose

Vehicle:

GEO SDO

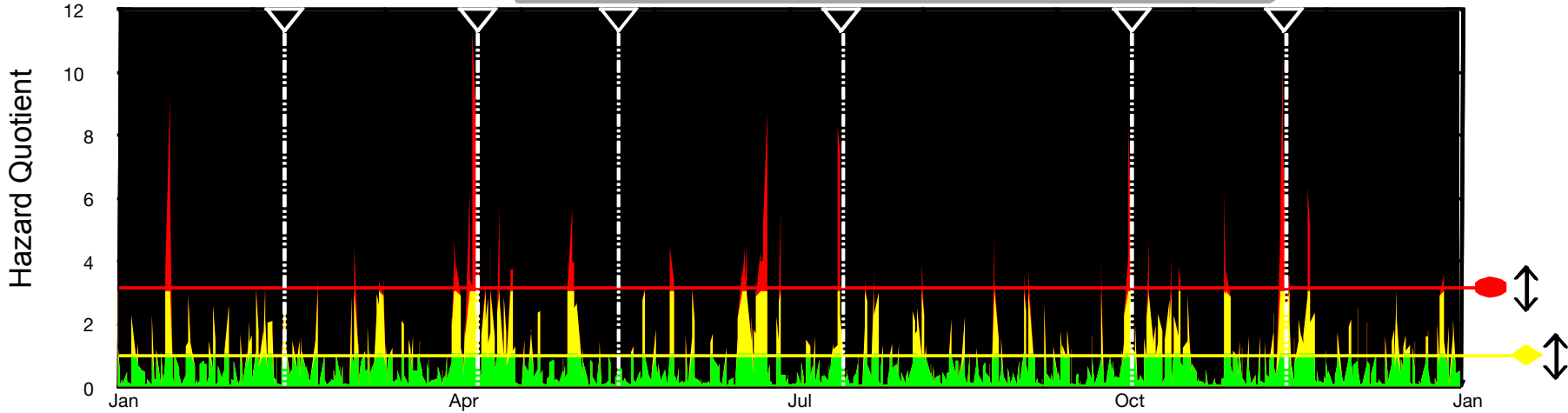
Anomaly Type:

VTCW RESET

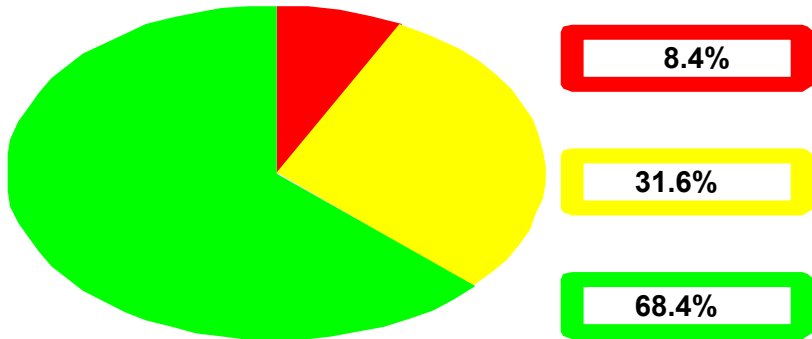
NEW

Hazard Indicator:

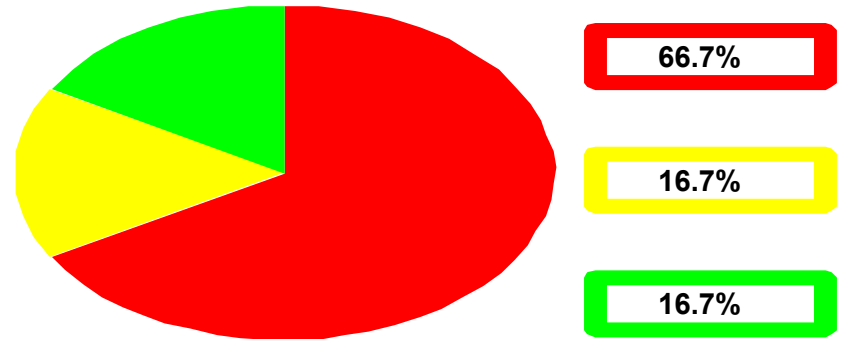
GOES >2MeV Electron Flux



% Time



% Anomalies



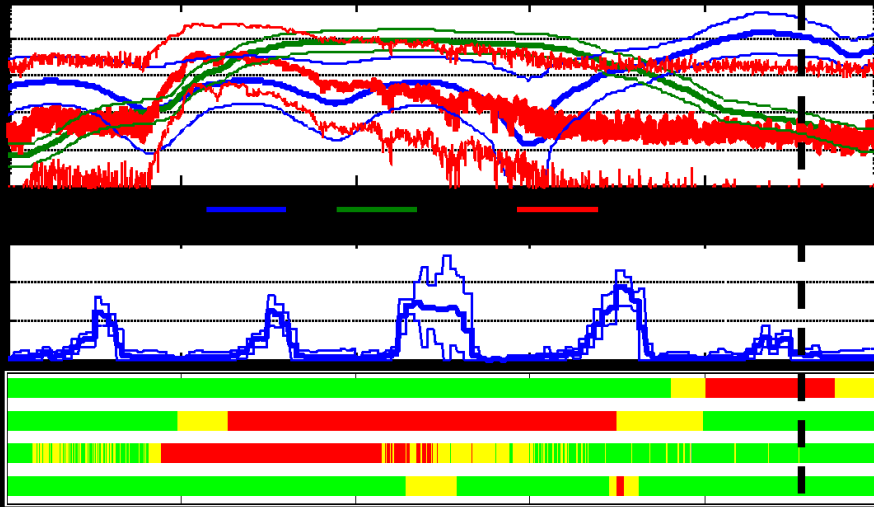
Expected anomalies per year for selected Hazard Indicator:

1

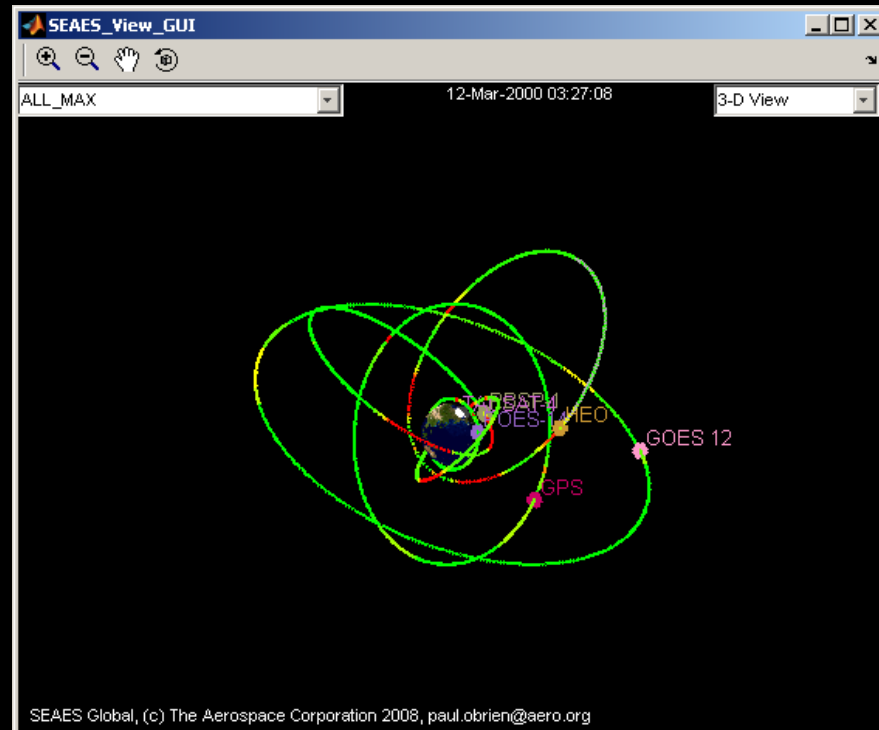
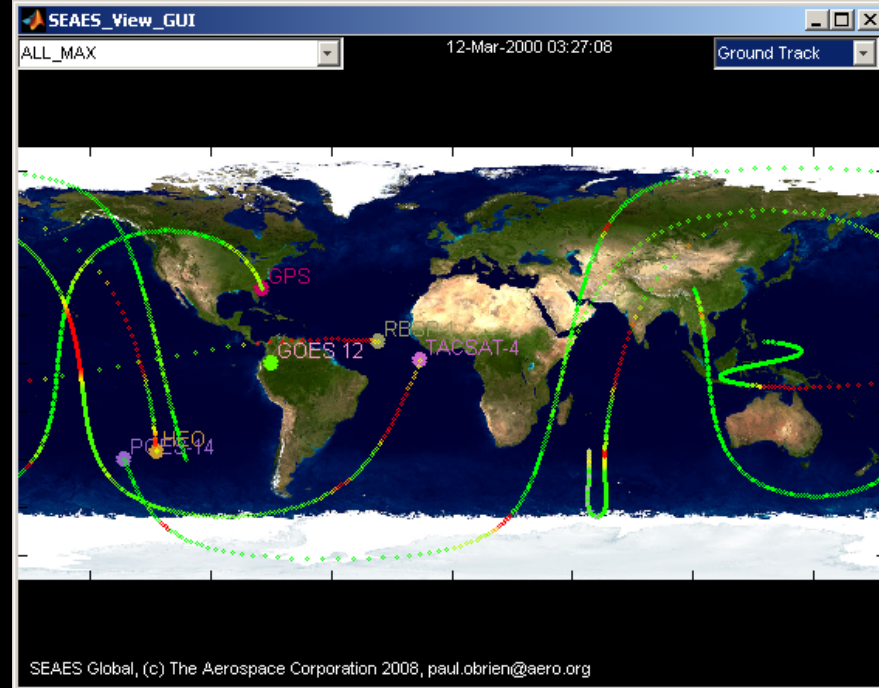
Suggested value based on selected Anomaly Type:

6

SAVE



More Sample Visuals from O'Brien



Future Goals / Possibilities

A satellite is shown in the foreground, oriented vertically. It features a large, silver, parabolic antenna at the top, a central body with various instruments, and two large solar panel arrays extending outwards. The background is a reddish, textured surface, likely representing Mars, with a bright horizon line at the bottom.

- Will be a product on Space Environment Automated Alerts & Anomaly Analysis Assistant (SEA⁵)

- Should be expanded to more orbits
 - MEO
 - LEO
 - HEO
 - Interplanetary



References & Thanks

Thanks to:

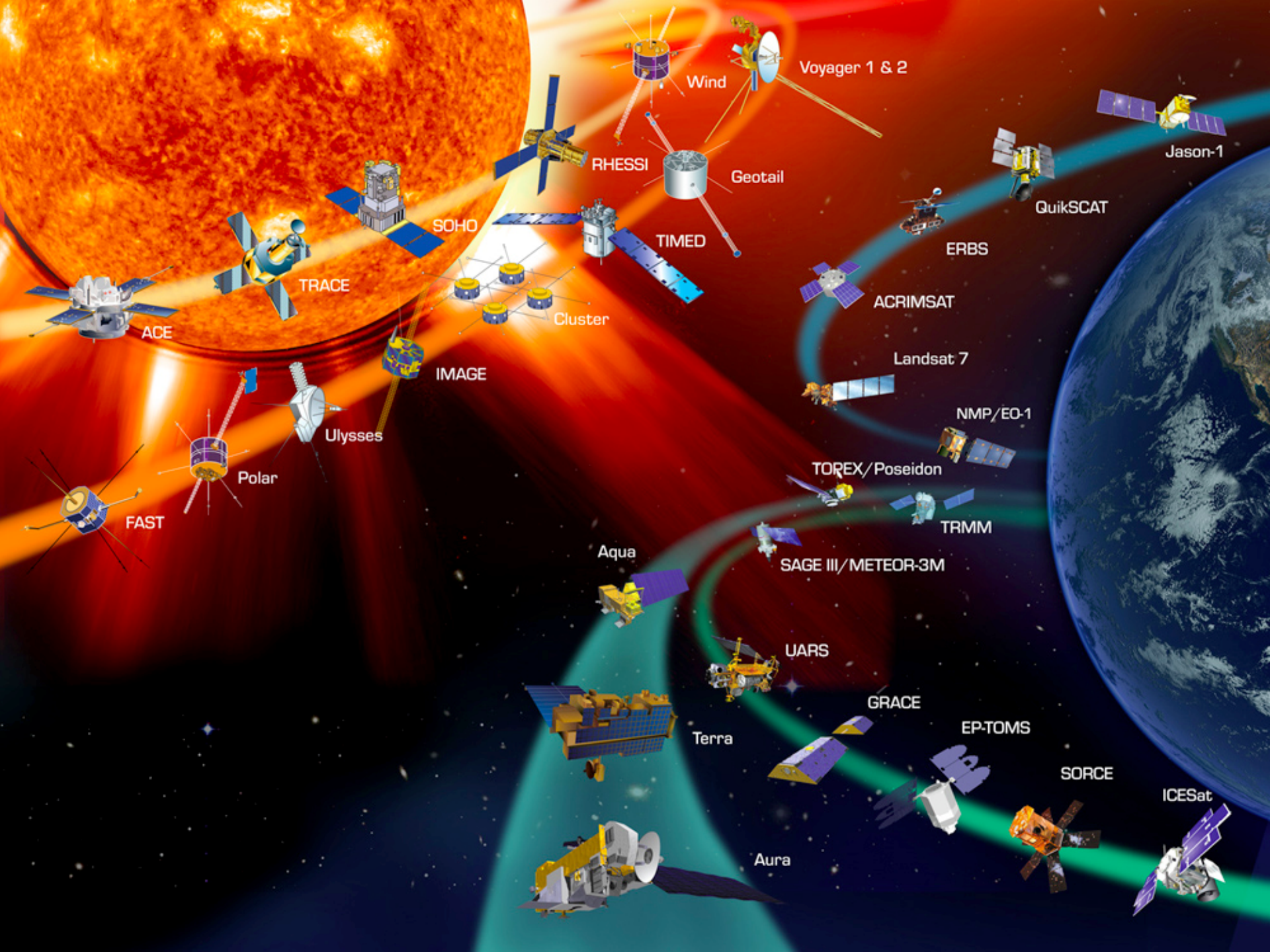
Yihua Zheng

Marlo Maddox

Tyler Schiewe

Paul O'Brien (2009)

Yi-Jiun Su (2014)



ACE

TRACE

SOHO

RHESSI

Wind

Voyager 1 & 2

Geotail

Jason-1

QuikSCAT

ERBS

TIMED

ACRIMSAT

Cluster

Landsat 7

IMAGE

NMP/E0-1

Ulysses

TOREX/Poseidon

Polar

TRMM

FAST

Aqua

SAGE III/METEOR-3M

UARS

Terra

GRACE

EP-TOMS

SORCE

ICESat

Aura